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THE UNIVERSITY OF TEXAS AT AUSTIN
Institute For Geophysics

EARTH STRAIN MEASUREMENTS
WITH THE TRANSPORTABLE LASER RANGING SYSTEM:
FIELD TECHNIQUES AND PLANNING

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ABSTRACT

We have conducted a feasibility study to examine the potential of the Transportable Laser Ranging System (TLRS) for monitoring the ground deformation around satellite ranging stations and other geodetic control points. Emphasis has been placed on testing the usefulness of the relative lateration technique. The temporal variation of the ratio of the length of each survey line to the mean length of all survey lines in a given area is directly related to the mean (h) ar strain rate for the area. The data from a series of experimental measurements taken over the Los Angeles basin from a TLRS station at Mt. Wilson show that such ratios can be determined to an accuracy of one part in 107 with a measurement program lasting for three days and without using any corrections for variations in atmospheric conditions. A numerical experiment using a set of hypothetical data indicates that reasonable estimates of the present shear strain rate and the direction of the principal axes in southern California can be deduced from such measurements over an interval of one to two years. Thus, the relative lateration from the TLRS appears to be a very economical way to monitor ground deformations, although there has been no opporunity yet to measure the actual ground strain by reoccupying the Mt. Wilson site.

Table of Contents

I.	Intro	oduction1
II.	TLRS	Ground-to-Ground Ranging
		Advantages and Problems3
		Relative Lateration4
		Relationship between Relative Lateration and Strain4
III.	Mt.	Wilson Experiment9
		Field Experiment9
		The Data12
		Range Ratios to a Single Reference Line12
		Time-of-Flight Ratios to the Mean
		An Alternative Atmospheric Correction
		A Test for Systematic Error Due to Atmospheric Conditions.17
		Results
IV.	Shea	r Strain Determination using Hypothetical Data21
v.	Conc	lusions and Recommendations
		Conclusions
		Recommendations
	Ackn	owledgements24
	Refe	rences
	Anno	andix

I. INTRODUCTION

With the recent development in ground-to-satellite laser ranging and Very Long Baseline Interferometry (VLBI) techniques, it is now possible to measure precisely distances between locations separated by several hundreds to thousands of kilometers. This makes it possible to monitor relative movements of globally distributed points on the earth for geodynamic studies. However, one question that must be answered is how representative each of these positions thus occupied is for the region in which it is located. If some of these locations are experiencing localized movements which are not representative of the region, the global measurement would give erroneous results. An answer to this questions can be found by measuring regional deformations around each location.

A conventional method for determining regional deformations is to perform repeated survey using an electro-optical distance measuring (EDM) device (e.g. Savage et al. [1981]). However, such surveys are expensive, and are rather limited in range. We, therefore, have looked for a better alternative. The development of the Transportable Laser Ranging System (TLRS) for ground-to-LAGEOS (Laser Geodynamics Earth Orbiting Satellite) ranging [Silverberg and Byrd, 1981] has given us an opportunity to test such an alternative. Because of its high sensitivity, being capable of detecting single photon returns, the TLRS can measure distances to small targets (retro-reflectors) at any visible points much beyond the normal ranges of other EDM devices. Thus, this system may provide economical measurements of strain fields in areas more than 200 km in diameter. If successful, such measurements will be valuable not only in the immediate neighborhood of satellite ranging stations, but also in understanding the dynamic behavior of both plate boundaries and areas internal to plates.

We have conducted a limited feasibility study to examine this potential. Although the TLRS is a powerful system, it also has certain limitations when used for a ground-to-ground ranging. The most important is the uncertainty of measurement results due to variability in atmospheric conditions. To bypass this problem and avoid the expense of flying an aircraft to monitor the atmospheric conditions along the path of the laser beam, we have examined the use of the relative lateration, or the ratio method, which was used earlier by Carter and Vincenty [1978] in an experimental survey around the McDonald Observatory.

We originally planned repeated field experiments at several sites in the western United States. However, because of many scheduling conflicts and delays associated with the overall TLRS-LAGEOS ranging experiments, the only field experiment we could perform during the current contract was a four-day measurement at Mt. Wilson over the Los Angeles basin in January, 1981. We have been unable to reoccupy this site for an actual strain measurement.

The present study, however, has given us some very encouraging results. Even with no atmospheric correction at all, the range ratios could be determined to an accuracy of one part in 10⁷. This is sufficient for an order-of-magnitude estimate of incremental shear strain in the southern California region if two measurements separated by one to two years are available. Higher accuracies would be attainable with repeated measurements.

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In this report, we first describe the advantages and problems of ground-to-ground ranging by a TLRS, leading to the use of relative lateration, or range-ratio method, and its relationship to the regional strain (Section II). Then, we present the data and analysis of the Mt. Wilson experiment (Section III). This is followed by a short treatment of regional strain determination using hypothetical data (Section IV). Finally, we present the conclusions from this feasibility study and offer some recommendations. Some pertinent data are presented in the Appendix.

II. TLRS GROUND-TO-GROUND RANGING

Advantanges and Problems

The TLRS is a highly mobile satellite laser ranging system designed to perform ground-to-LAGEOS range measurements. It is also highly sensitive, being capable of determining the range to a LAGEOS satellite with return signals as low as one photelectron every 20 to 50 laser shots [Silverberg et al. 1982]. Used as a ground-to-ground ranging device, it can measure the distance to any single 1 inch (25 mm) corner reflector within sight at very low laser power level. The measureable range is limited only by the curvature of the earth. The required power level is so low that, unlike some systems used for similar measurements, the laser beam can be maintained many orders of magnitude below the eye-damage threshold.

The practical precision of the TLRS range measurements is limited to about 1.5 cm for a one minute average, which is somewhat worse than those of conventional EDM devices using modulated laser beams. However, the long range capability of the TLRS reduces the relative error to well within the limits of interest in conventional surveys. The TLRS has an automatic pointing system, an automatic calibration system and other features which lend themselves to providing many horizontal (ground-to-ground) line measurements on an operational basis. Thus it will be a good device to use if the data it provides is sufficient to determine the regional deformation at a high-enough accuracy.

The most serious problem in using the TLRS for ground-to-ground ranging is the atmospheric effect. The temperature, pressure, and to a lesser degree water vapor influence the index of refraction of air, and thus the speed of a laser beam through the atmosphere. To obtain the absolute distance between two points from a time of flight measurement through air, one must make corrections for these atmospheric variables.

Estimates of these atmospheric variables along the beam path may be made based on measurements at the two end points. This, however, is unsatisfactory for long lines. A more precise way is to measure directly the atmospheric condition along the beam path by flying an aircraft during the ranging. This, though done in practice, is a costly operation. A third alternative is to use more than one wavelength for ranging. Using the dispersive characteristics of light in air, one can correct for the atmospheric effects [Huggett, et al. 1977].

The present TLRS operates in a single color. Flying an aircraft, we judged, is too costly for repeated measurements in many directions. Thus we had to look for another alternative.

Relative Lateration

One way to improve the accuracy of range measurements without relying on expensive in-flight measurement of atmospheric conditions is to use a relative lateration technique, or the "ratio method" (Robertson, 1972). Instead of attempting to measure the absolute length of each survey line to high accuracy, this technique determines only the ratios of distances. This method is based on a supposition that the temporal changes of atmospheric conditions along several survey lines within a given region are similar to each other. Therefore, even when the time of flight of a laser beam in each line fluctuates with changing atmospheric conditions, the ratios of the times of flight along different survey lines tend to vary little with time.

Carter and Vincenty [1978] used this method in an experimental EDM survey around the McDonald Observatory in 1977. They obtained sets of measurements, one month apart, consistent to one to two parts in 107. They have just repeated this experiment and the data is now being analyzed. Since the results of Carter and Vincenty appear to be quite promising, we have decided to try the same for our TLRS measurements.

Relationship Between Relative Lateration and Strain

Unlike absolute measurements of distances, the relative distance measurements repeated after a certain time period will not give all of the components of deformation, or incremental strain, for the time period unless at least one survey line is measured absolutely. However, a clear relationship exists between the changes of relative distances and incremental shear strain.

Let us consider n survey lines radiating from a central station. In the present case, the TLRS is located at the central station and a retroreflector is located at the end of each radiating line. Assume that all lines lie in a horizontal plane, neglecting both the curvature of the earth's surface and topographic height differences. Choosing a coordinate system with the origin at the central station, positive x towards east and positive y towards north, the original length of line i to the reflector at coordinates (x_i, y_i) at the time of the initial survey is given by

$$s_{i} = (x_{i}^{2} + y_{i}^{2})^{\frac{1}{2}}$$
 (1)

Now assume that between the initial survey and a subsequent survey the entire area of the survey undergoes a uniform deformation represented by incremental strain components ε_{xx} , ε_{xy} and ε_{yy} . Then, the line length becomes

$$\mathbf{s_{i}'} = [(\mathbf{x_{i}} + \varepsilon_{\mathbf{x}\mathbf{x}}\mathbf{x_{i}} + \varepsilon_{\mathbf{x}\mathbf{y}}\mathbf{y_{i}})^{2} + (\mathbf{y_{i}} + \varepsilon_{\mathbf{x}\mathbf{y}}\mathbf{x_{i}} + \varepsilon_{\mathbf{y}\mathbf{y}}\mathbf{y_{i}})^{2}]^{\frac{1}{2}}$$

$$= [\mathbf{x_{i}}^{2} + \mathbf{y_{i}}^{2} + 2\varepsilon_{\mathbf{x}\mathbf{x}}\mathbf{x_{i}}^{2} + 4\varepsilon_{\mathbf{x}\mathbf{y}}\mathbf{x_{i}}\mathbf{y_{i}} + 2\varepsilon_{\mathbf{y}\mathbf{y}}\mathbf{y_{i}}^{2}]^{\frac{1}{2}}$$

$$= \mathbf{s_{i}}[1 + 2\varepsilon_{\mathbf{x}\mathbf{x}}\sin^{2}\alpha_{i} + 4\varepsilon_{\mathbf{x}\mathbf{y}}\sin\alpha_{i}\cos\alpha_{i} + 2\varepsilon_{\mathbf{y}\mathbf{y}}\cos^{2}\alpha_{i}]^{\frac{1}{2}}$$

$$= \mathbf{s_{i}}[1 + \varepsilon_{\mathbf{x}\mathbf{x}}\sin^{2}\alpha_{i} + \varepsilon_{\mathbf{x}\mathbf{y}}\sin\alpha_{i} + \varepsilon_{\mathbf{y}\mathbf{y}}\cos^{2}\alpha_{i}] \qquad (2)$$

where $\alpha_i = \tan^{-1}(x_i/y_i)$ is the azimuth of the line i measured clockwise from north, and the higher order terms in strain have been neglected. Then, the range increment δ_i is given by

$$\delta_{i} = s_{i}' - s_{i} = s_{i} \left[\varepsilon_{xx} \sin^{2} \alpha_{i} + \varepsilon_{xy} \sin 2\alpha_{i} + \varepsilon_{yy} \cos^{2} \alpha_{i} \right]$$
 (3)

Next define original mean range and range ratios to the mean, respective, as

$$\overline{s} = \sum_{i=1}^{n} s_i/n \tag{4}$$

and

$$r_i = s_i/\bar{s} \tag{5}$$

Then, the subsequent mean range and range ratios are

$$\overline{s}' = \sum_{i=1}^{n} s_i'/n = \overline{s} + \sum_{i=1}^{n} \delta_i/n$$
 (6)

and

$$r_{i}' = s_{i}'/\overline{s}' = (s_{i} + \delta_{i})/(\overline{s} + \sum_{i=1}^{n} \delta_{i}/n)$$

$$= r_{i}(1 + \delta_{i}/s_{i} - \sum_{i=1}^{n} \delta_{i}/n\overline{s})$$
(7)

where the higher order terms are again neglected. The increment of the range ratio is, therefore,

$$r_{i}' - r_{i} = r_{i} (\delta_{i}/s_{i} - \sum_{i=1}^{n} \delta_{i}/ns)$$
 (8)

Then, range ratio increment normalized by the original range ratio is given by

$$\gamma_{i} = (r_{i}' - r_{i})/r_{i} = \delta_{i}/s_{i} - \sum_{i=1}^{n} \delta_{i}/n\overline{s}$$
(9)

Substituting eq. (3) into eq. (9), and using (5), we obtain

$$\gamma_{i} = \left[\sin^{2}\alpha_{i} - \sum_{i=1}^{n} (r_{i}\sin^{2}\alpha_{i})/n\right] \varepsilon_{yy} \\
+ \left[\sin^{2}\alpha_{i} - \sum_{i=1}^{n} (r_{i}\sin^{2}\alpha_{i})/n\right] \varepsilon_{xy} \\
+ \left[\cos^{2}\alpha_{i} - \sum_{i=1}^{n} (r_{i}\cos^{2}\alpha_{i})/n\right] \varepsilon_{yy} \tag{10}$$

Equation (10) may give one an impression that a set of measurements of the normalized range ratio increments γ_i would give the incremental strain components ϵ_{xx} , ϵ_{xy} and ϵ_{yy} . However, this impression is incorrect because the coefficients of ϵ_{xx} and ϵ_{yy} are not independent of each other, as their sum vanishes, and therefore ϵ_{xx} and ϵ_{yy} cannot be determined uniquely.

Now let

$$\Theta = \varepsilon_{\mathbf{x}\mathbf{x}} + \varepsilon_{\mathbf{y}\mathbf{y}} \tag{11}$$

and

$$\Psi = \epsilon_{xx} - \epsilon_{yy} \tag{12}$$

Then,

$$\varepsilon_{xx} = \frac{1}{2} (\Theta + \Psi) \tag{13}$$

and

$$\varepsilon_{yy} = \frac{1}{2} (\Theta - \Psi) \tag{14}$$

Substituting (13) and (14) into (10), we obtain

$$\gamma_{i} = \left[\sin 2\alpha_{i} - \sum_{i=1}^{n} (r_{i} \sin 2\alpha_{i})/n\right] \varepsilon_{xy}$$

$$-\frac{1}{2} \left[\cos 2\alpha_{i} - \sum_{i=1}^{n} (r_{i} \cos 2\alpha_{i})/n\right] \Psi$$
(15)

The coefficients of ϵ_{xy} and Ψ are known quantities for the initial setup of the survey lines. Thus, for a set of measurements of the normalized range ratio increments γ_i , the incremental shear strain components ϵ_{xy} and Ψ can be determined by a least-square inversion of eq. (15).

Finally, the maximum incremental shear strain S and the direction of the principal strain axes β are given by

$$s = [(2\varepsilon_{xy})^2 + \Psi^2]^{\frac{1}{2}}$$
 (16)

and

$$\beta = \frac{1}{2} \tan^{-1} (2\varepsilon_{xy}/\Psi) \tag{17}$$

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The dilation Θ of eq. (11) disappears from eq. (15), and thus cannot be determined. This is expected because any uniform compression or expansion of the entire area causes no change in range ratios.

The treatment above assumes uniform deformation of the entire region. If for some reason, such as the existence of active faults within the area, the regional deformation is not uniform, large residuals will show up in the least-square inversion of eq. (15). Thus, any residuals significantly larger than the measurement errors will indicate heterogeneous strain.

The remaining question is how accurately we can estimate the normalized range ratio increments γ_i . Since the measurements are done in terms of time of flight of light beams, the uncertainty in speed of light is the determining factor. The average speed of light, c_i , between the central station and a reflector i may be expressed as the sum of four components:

$$c_{i} = c_{o} + \ell_{i} + w_{c} + w_{i} \tag{18}$$

where $c_{\rm O}$ is the speed of light in standard air, which is constant for all survey lines at all times; $\ell_{\rm i}$ is the correction attributable to the reflector location, which is time invariant for a given reflector; $w_{\rm C}$ is a component of correction attributable to weather common to all reflectors at a given time; and $w_{\rm i}$ is the residual weather correction. The line length $s_{\rm i}$ is given in terms of round-trip time of flight, $t_{\rm i}$, as

$$s_{i} = \frac{1}{2}(c_{0} + \ell_{i} + w_{0} + w_{i})t_{i}$$
 (19)

the mean range as

$$\bar{s} = \frac{1}{2} (c_0 \bar{t} + \sum_{i=1}^{n} \ell_i t_i / n + w_c \bar{t} + \sum_{i=1}^{n} w_i t_i / n)$$
 (20)

where $\bar{t} = \sum_{i=1}^{n} t_i/n$ is the mean time of flight, and the range ratio to the mean

$$r_{i} = u_{i}[1 + (l_{i} - \sum_{i=1}^{n} l_{i}t_{i}/n\bar{t} + w_{i} - \sum_{i=1}^{n} w_{i}t_{i}/n\bar{t})/c_{o}]$$
 (21)

where $u_i = t_i/\bar{t}$ is the time-of-flight ratio to the mean, and the higher-order terms have been neglected. Finally, the normalized range ratio increment is given as

$$\gamma_{i} = \eta_{i} + [(w_{i}' - w_{i}) - \sum_{i=1}^{n} (w_{i}' - w_{i})t_{i}/n\bar{t}]/c_{o}$$
 (22)

where $\eta_i = (u_i^{\ i} - u_i^{\ i})/u_i$ is the normalized time-of-flight ratio increment and quantities with primes designate those at subsequent measurement as before. The higher-order terms are again neglected. Note that the common weather component, w , is eliminated by taking the range ratio (21), and the location specific components, ℓ_i 's, are eliminated by normalization (22), leaving only the residual weather components w_i 's.

The normalized time-of-flight ratio increment, $\eta_{.}$, thus approximates the range ratio increments, $\gamma_{.}$, with a small error due to residual weather term. The latter is not location specific, and is not common to all lines at a given time. If this term is sufficiently small, then we can substitute η_{i} for γ_{i} in calculating the shear strain increment using (15).

III. MT. WILSON EXPERIMENT

Field Experiment

At the request of the NASA Crustal Dynamics Project, the TLRS team from the McDonald Observatory of the University of Texas, led by Dr. Eric Silverberg, deployed the TLRS at Mt. Wilson, California, in January of 1981. At the same time, two of us (H.J.D. and T.C.) scouted the surrounding area for suitable target sites and selected the reflector locations. Then, a field party from the National Geodetic Survey (NGS), which was dispatched at the request of NASA to help us, deployed retroreflectors at the chosen sites. The survey lines selected for the site are shown in Figure 1. Table 1 lists the nominal coordinates of the base station (TLRS (199)) at Mt. Wilson and of the end points of the lines, where the retroreflectors were installed. Also listed in Table 1 are the approximate look angles from Mt. Wilson and ranges as computed from the indicated coordinates using the IAG standard ellipsoid Geodetic Reference System 1967.

Each reflector except the one at Cahuenga was a metal box containing an array of three $1\frac{1}{2}$ inch (38mm) corner cubes, supplied by the NGS. The box was mounted on a tripod and placed directly over the station mark using an optical blumb bob. This elaborate configuration made it necessary to guard the reflector continuously for the entire duration of the experiment. The reflector used at Cahuenga was designed by one of us $(\mathfrak{T}_*\mathcal{C}_*)$ for unmanned operation. It contained a single 1 inch (25mm) corner cube and was fastened to an outcrop with anchor bolts at a site off the station mark, thus concealed from public view.

The reference point of the TLRS, from which the raw time-of-flight measurements were made, was slightly offset from the Mt. Wilson station mark given in Table 1. The measured coordinates of the station mark relative to the TLRS were:

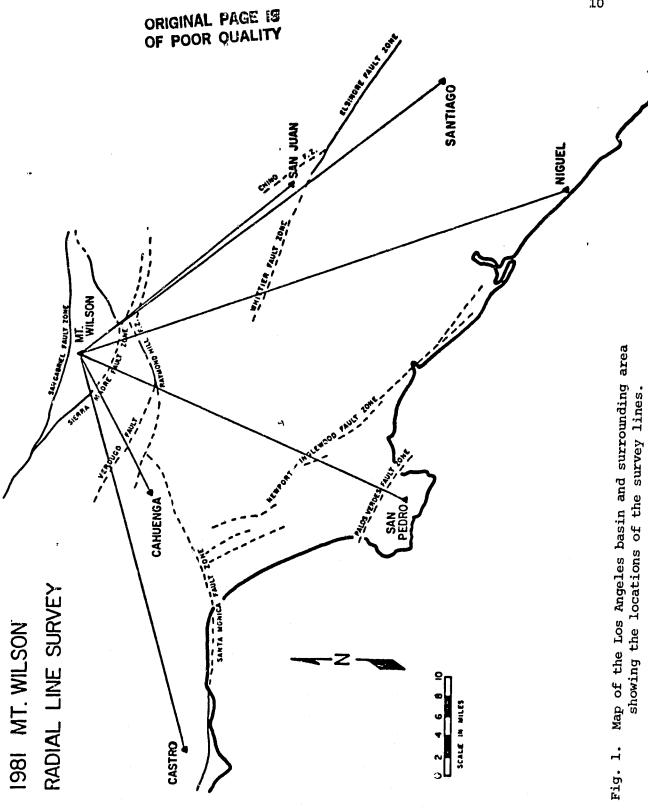
x = -1.4873 m (west)

y = 0.5093 m (north)

z = -3.3709 m (below)

The resulting corrections, to be applied to the observed quantities to reduce them to the reference mark, are listed in Table 2. The corrections can be applied at any stage of data reduction.

After the initial setup, which began on January 9, 1981, the horizontal ranging data were collected over the four-day interval January 23 through 26, 1981, in cooperation with the NOAA National Geodetic Survey. Each of the reflector sites except Cahuenga was manned continuously during the entire experiment to record the temperature, pressure and relative humidity at the site at about 30 minute intervals. The details of the data acquisition are given in Silverberg et al. [1982].



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Table 1. Stations Used in Mt. Wilson Line Survey

Station	Longitude	Latitude	Elevatio m		Angle Altitude	Range m	Note
Mt. Wilson	241° 56' 17.85"	34°13′21.58"	1722.0				1
Castro	241° 12′55.40"	34'05'08.57"	858.9	257.366	-1.030	68393	2
Cahuenga	241° 40′29.81"	34° 08′ 13.00 "	554.1	248.687	-2.681	26104	3
San Pedro	241° 39′55.73"	33° 44′ 40.88"	447.1	205.506	-1.508	58729	4
Niguel	242° 16′00.18"	33° 30′ 44.03"	288.0	158.814	-1.353	84459	
Santiago	242 28 00 . 10"	33° 42′ 37.89"	1733.0	139.168	-0.329	74933	
San Juan	242*15'45.99"	33° 54′ 49 . 47 "	543.0	138.751	-1.688	45536	

¹ New marker 9.408m from Mt. Wilson E10A @345°17'

Table 2. Corrections to be Applied to Observations to Reduce to Mt. Wilson Gound Marker

	Round-Trip	۔ مد ندر چنز مید سن نیدر پیدر سن کا کہ	Range	Ratio*
Station	Time of Flight	Range	(1)	(2)
	ns	m 	ი	pm
Castro	-9.344	-1.4003	-25.08	-26,78
Cahuenga	-9.054	-1.3568	-23.35	-23.51
San Pedro	-1.798	-0.2694	-5.91	-7.93
Niguel	6.222	0.9325	13.61	9.93
Santiago	8.931	1.3385	20.64	17.07
San Juan	8.432	1.2636	20.09	17.64

^{* (1)} Ratio to mean range

² Solitice Canyon B2 Aux. 1, which is 14.107m FNE of Castro 1898 3 Reference mark #3 of Cahuenga #2, 13.329m @257°47′ from Cahuenga #2 4 L7 Ecc. San Pedro Hills, which is 12.576m @318°57′ from San Pedro #3

⁽²⁾ Same but excluding Cahuenga and Niguel

The Data

The raw field data were initially processed at the University of Texas at Austin by the McDonald Observatory group. As described in detail by Silverberg et al. [1982], the processing of the raw data involved accumulation of individual photon returns into 200 psec bins, smoothing of the coadded returns by three-bin (600 psec) running averages, cross-corretation with a reference standard to eliminate long-term drift in the calibration constants, adjustments to account for certain measurement irregularities, and removal of a 86.8 nsec constant calibration correction.

The calibrated round-trip time-of-flight data, shown in Figure 2 and listed in Table Al in the Appendix, have not been corrected for the offset of the TLRS from the ground marker (Table 2). The data for Cahuenga were not used for the analysis because of certain processing difficulties encountered for the data for this station.

The data gap during the second day of observation was due to an interruption in data acquisition caused by rain which accompanied the passage of a cold front. The meteorological data taken at Mt. Wilson site and other stations are shown in Figures Al through A3, and are listed in Table A2 in the Appendix.

As expected, the raw time-of-flight data show large fluctuations, which are only partially correlated with the meteorological data. The relative RMS deviations of the time-of-flight data (Table 3, colume 3) range from 1.53 ppm for Niguel, which was surveyed only after the passage of the cold front, to 3.78 ppm for San Juan, which was the shortest line. The weighted average for all lines is 2.84 ppm.

Range Ratios to a Single Reference Line

Silverberg et al. [1982] calculated the time-of-flight ratios and atmosphere-corrected range ratios to a reference line following the procedure used by Carter and Vincenty [1978]. The reference line they chose was a smoothed curve (a cubic spline) through the Santiago data. Their results (Table 3, column 5) show relative RMS deviations of time-of-flight ratios ranging from 0.4 ppm for Niguel to 1.6 ppm for San Juan. The weighted average for all lines is 1.0 ppm, which is about a factor of three improvement from the fluctuation of the time-of-flight data.

Their results for the range ratios with atmospheric corrections based on end-point meteorological data did not fare as well. In fact the relative RMS deviations increased typically about 40% from those of uncorrected time-of-flight ratios [Silverberg et al. 1982].

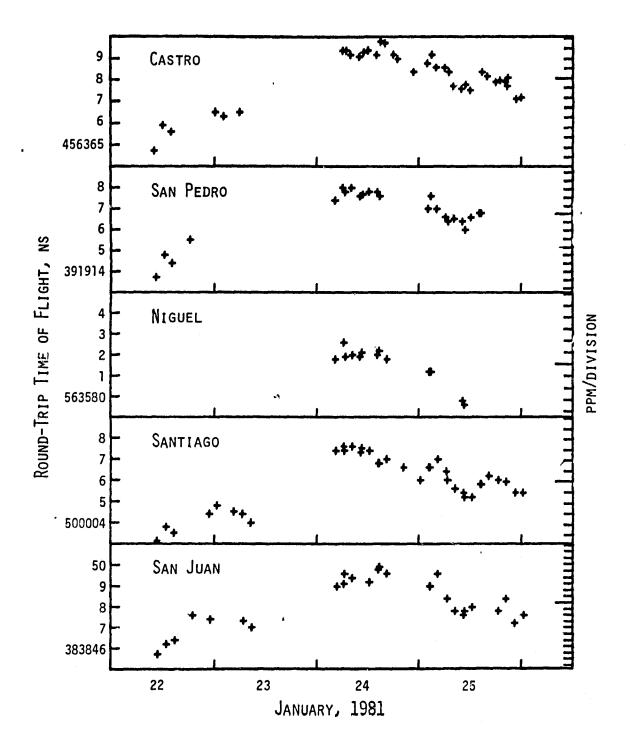


Fig. 2. Round-trip time of flight from Mt. Wilson. The data are not corrected for the marker offset.

The main reason for the poor performance of atmosphere-corrected values is the difficulty of making proper atmospheric corrections. A comparison of the variations of the group index of refraction calculated from the temperature and pressure at end points (Figure 3; also listed in Table A3 in the Appendix) with the time-of-flight variations (Figure 2) clearly shows that long-term variations are fairly well matched but shorter diurnal fluctuations are larger for the index of refractions than for the times-of-flight. Thus the index-of-refraction correction per Carter and Vincenty [1978] over-compensates for diurnal variations.

Time-of-Flight Ratios to the Mean

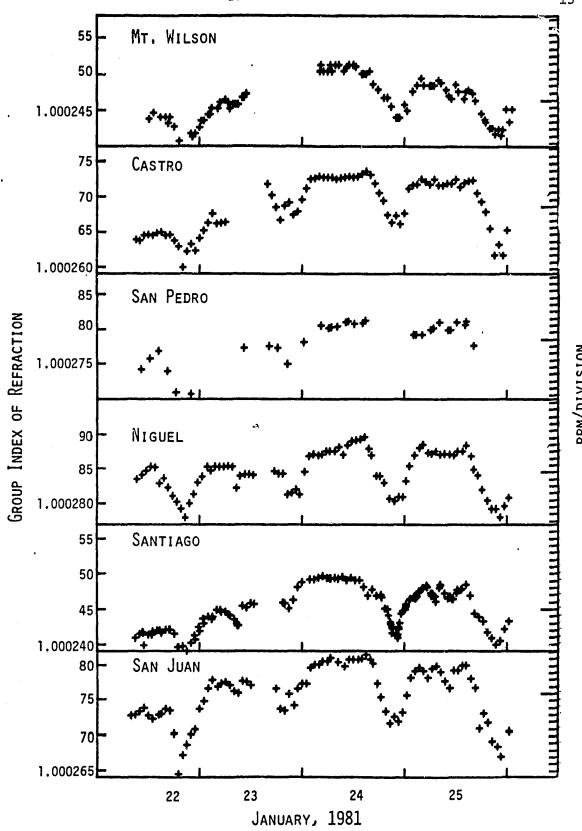
In order to be consistent with the range-ratio/strain relationship of the preceding section, we calculated the time-of-flight ratios to the mean. Since the time-of-flight measurements to all targets were not made exactly simultaneously, the data were linearly interpolated before the mean time-of-flight for a given time was calculated. (Higher order interpolations or a spline approximation might be better, but we judged the difference would be small.) Also, since we had data for Niguel only during the last half of the experiment, this station was excluded from the mean time-of-flight calculation.

The resulting time-of-flight ratios to the mean (Figure 4; also listed in Table A4 in the Appendix) show a further improvement in the fluctuations of the results. The relative RMS deviations (Table 3, column 5) now range from 0.36 ppm for Niguel to 1.24 ppm for San Juan, with the weighted average of 0.71 ppm for all lines, a factor of four improvement from the raw time-of-flight data.

An Alternative Atmospheric Correction

As stated earlier, the short-term, diurnal fluctuations in the index of refraction at end points exceed the observed fluctuations in the time-of-flight values. This is probably due to the larger fluctuation of the atmospheric temperature near the ground than those in most of the intervening air mass; a result of the base station and most of the target stations being located well above the intervening terrain. In this situation, a standard correction procedure like that of Carter and Vincenty [1978] is not really applicable, and some alternate procedures are needed.

An experimental procedure we tried was to estimate the average temperature of the air mass by low-pass filtering the mean of the temperatures measured at the end points. The filter we used was a simple one of adding all previous temperature readings each weighted by a factor proportional to a negative exponential of the elapsed time. After



Fgi. 3. Group index of refraction computed from atmospheric data at end points.

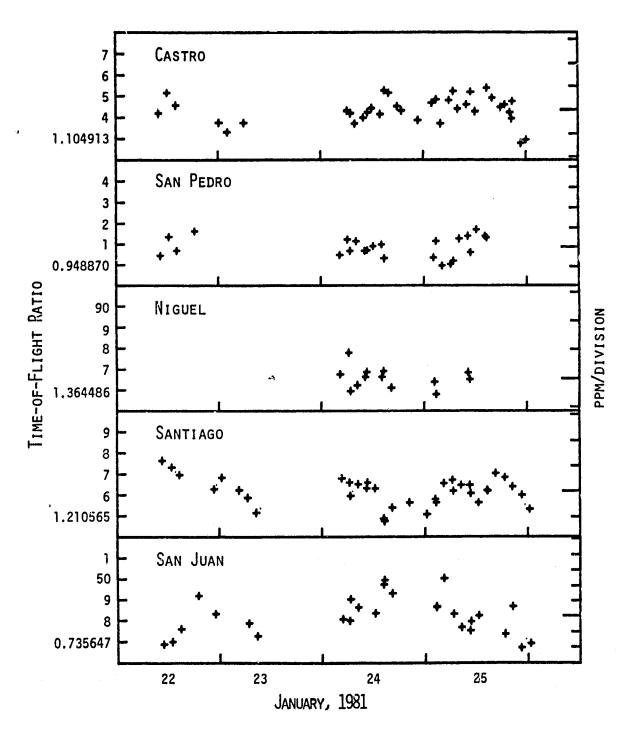


Fig. 4. Time-of-flight ratios to the mean.

trials with such filters of several different time constants, a time constant of 12 hours was found to give the best result. The resulting relative RMS deviations of the ranges thus corrected for atmospheric conditions (Table 3, column 4) range from 0.96 ppm for Castro to 2.05 ppm for San Juan with the weighted average of 1.38 ppm for all lines. This is about a factor of two improvement from the raw time-of-flight data. However, the range ratios calculated from these corrected ranges do not show any significant improvement over those of the uncorrected ratios. The relative RMS deviations of the corrected range ratios (Table 3, last column) range from 0.52 ppm for Niguel and Santiago to 1.21 ppm for San Juan, with the weighted average of 0.72 ppm for all lines.

A comparison of the relative RMS deviations of various quantities in Table 3 reveals that the best result is obtained for the uncorrected time-of-flight ratios to the mean. The atmospheric corrections did not improve the RMS deviations at all when ratios were taken.

A Test for Systematic Error Due to Atmospheric Conditions

The reliability of the relative lateration depends on the validity of the assumption that the temporal changes of atmospheric conditions are similar for all survey lines in the area so that their effects cancel out when ratios are taken. If this assumption is incorrect, a systematic error due to varying atmospheric conditions is introduced into the measured time-of-flight ratios. The greatly different atmospheric conditions before and after the passage of a cold front during the experiment gave us an opportunity to test this assumption.

The test we performed is the likelihood ratio test. We divided the time-of-flight ratios of Table A4 for each line into two subsets, the first half and the last half, of equal size (the last half was one greater than the first half if the total number was odd). If the systematic error due to atmospheric conditions is significantly large, the mean ratio, μ_1 , for the first subset will be significantly different from that, μ_2 , for the second subset. Setting up a null hypothesis $\mu_1 = \mu_2$, if it is true, then the likelihood ratio statistic

$$t = [n_1 n_2 / (n_1 + n_2)]^{\frac{1}{2}} (\mu_1 - \mu_2) / [(n_1 \sigma_1^2 + n_2 \sigma_2^2) / (n_1 + n_2 - 2)]^{\frac{1}{2}}$$
(23)

has a t distribution with n_1+n_2 -2 degrees of freedom, where n_1 and n_2 are the sample sizes of the two subsets and μ_1,μ_2 , σ_1^2 and σ_2^2 are used to designate the sample means and the sample variances of the first and the second subsets, respectively, for convenience.

At 90% significance level, the t distribution has values of 1.69 for 34 degrees of freedom and 1.80 for 11 degrees of freedom, while the t values computed from the data, Table 4, are much smaller. Therefore, the null hypothesis cannot be rejected at this level of significance.

ORIGINAL PAGE 19 OF POOR QUALITY

Table 3. Comparison of Relative RMS Deviations in ppm

Stat .ion	Number of Data Points	Uncorrected Time of Flight	Corrected Range (a)	Uncorrected T-o-F Ratio to Santiago(b)		Corrected Range Ratio to the mean
Castro	36	2.68	0.96	0.7	0.56	0.54
San Pedro	24	2.91	1.12	0.8	0.52	0.54
Niguel	13	1.53	1.81	0.4	0.36	0.52
Santiago	36	2.45	1.03	(0.2)	0.54	0.52
San Juan	27	3.78	2.05	1.6	1.24	1.21
A11	136	2.84	1.38	1.0	0.71	0.72

Table 4. Likelihood Ratio Test for Non-equality of Means

Station		Subset	1		Subset	2	Degree	
	n ₁	μ_1	$\sigma_{\mathbf{l}}$	n ₂	μ_2	σ ₂	Freedom	
Castro	18	1.104914282	0.000000513	18	1.104914438	0.000000694	34	0.745
San Pedro	12	0.948870911	0.000000356	12	0.948970818	0.000000598	22	0.445
Niguel	6	1.364486713	0.000000572	7	1.364486460	0.000000365	11	0.889
Santiago	18	1.210566249	0.000000774	18	1.210566130	0.000000515	34	0.531
San Juan	13	0.735648159	0.00000030	14	0.735649288	0.000000983	25	0.353

⁽a) Corrected by using 12-hour low-pass filtered temperature(b) From Silverberg et al. [1982]. The deviation for Santiago is from the smoothed curve, and is not included in calculating the average for all stations.

In other words, no significant difference is found between the mean time-of-flight ratios in the first and second halves of the experiment for any of the lines surveyed.

Results

Since there is no evidence for systematic errors caused by atmospheric conditions, the most likely estimates of the mean time-of-flight ratios and their variances (and standard deviations) can be calculated from the entire data set. The results are shown in Table 5. Also listed in this table are the mean time-of-flight and the mean distances. The latter were calculated using atmospheric corrections based on the low-pass filtered temperatures described earlier and pressures interpolated to the average height of the beam from the end-point measurements (extrapolation in case of Santiago because the average height of the beam was lower than either end point). A group index of refraction of n =1.00028975 at the wavelength of 0.5320 µm, calculated from the formula given in American Institute of Physics Handbook [1972, p. 6-111] for standard dry air with 0.03% carbon dioxide at 15°C and 760 mm Hg, is used. No other corrections have been applied to the calculated distances; thus they are subject to minor systematic errors.

The estimated relative standard deviations of the mean time-of-flight ratios are approximately 1 x 10⁻⁷ except for San Juan, which is the shortest line. In comparison, Savage and Prescott [1973] estimate that the standard deviation of their Geodolite measurements of distances are 3 and 8 mm for lengths of 1 and 37 km, respectively. Thus, the precision of the present time-of-flight ratios is at least a factor of two better than that of their distance measurements. Furthermore, their distances had to be corrected for temperature and humidity readings made with an aircraft flying along the line of sight, while the present time-of-flight ratios required no atmospheric correction at all.

Multiwavelength measurements of distances are definitely better than the above two in terms of relative accuracy. Huggett and Slater [1975] and Slater and Huggett [1976] show the standard deviation of individual distance measurements to be less than 1 x 10^{-7} on a 10.1 km line. By taking the mean of many measurements, which is practical in this case, the accuracy can be improved further. The ranges attainable with the multiwavelength system, however, are quite limited compared with the TLRS measurements.

ORIGINAL PAGE 18 OF POOR QUALITY

Table 5. Mean Time of Flight, Distance and Ratios*

Station	Mean Time of Flight ns	Mean Distance m	Mean .T-o-F Ratio	S.D, of Mean T-o-F Ratio	Relative S.D. ppm
Castro	456358.75	68388.52	1.10488758	9.00000010	0.09
San Pedro	391914.94	58730.88	0.94886293	0.00000010	0.11
Niguel	563587.77	04456.72		0.00000014	0.10
Sant Lago	500014.83	74931.62		0.00000011	0.09
San Juan	303856.63	45535.87	0.73566587	0.00000018	0.24

^{*} These results have been corrected for the TLRS/ground-marker offset.

Table 6. Increment in Normalized Ratio Due to Hypothetical Strain Increment and Rounded Values for Testing

Station	Normalized Ratio Incremer ppm	Rounded to 0.1 ppm
Castro	0.052	Ø. 1
San Pedro	-0.158	-0.2
Niguel	-0.030	0.0
Santiago	0.068	0.1
San Juan	0.070	0.1

SHEAR STRAIN DETERMINATION USING HYPOTHETICAL DATA

We have been unable to reoccupy the Mt. Wilson site for a repeat measurement, which would allow a testing of the ratio method for a shear strain determination in the region. This section, therefore, describes an exercise we have conducted to see how well we can determine the regional shear-strain increment using a set of hypothetical data.

We assume a hypothetical strain increment described by

 $\epsilon_1 = 0.1 \times 10^{-6}$: maximum extension $\epsilon_2 = -0.2 \times 10^{-6}$: maximum compression $\epsilon_1 - \epsilon_2 = 0.3 \times 10^{-6}$: maximum shear $\beta = 110^{\circ}$.

azimuth of maximum positive principal axis (extension) measured clockwise

from north

This strain increment is approximately the annual strain increment in southern California observed by Savage et al. [1981]. The resulting increments in normalized range ratios for the five survey lines used in the Mt. Wilson experiment are listed in the center column of Table 6. Since we will not be able to measure these ratio increments at this accuracy, we use the values rounded to 1×10^{-7} , as given in the rightmost column of Table 6.

Substituting these rounded ratio increments into eq. (15), and inverting it in a least-squares sense, we obtain the following results:

$$\varepsilon_1 - \varepsilon_2 = 0.40 \times 10^{-6}$$
 $\beta = 109.5^{\circ}$

The result describes the original hypothetical shear-strain increment reasonably well. A trial with a rounding to 1 x 10 8 results in almost complete duplication of the hypothetical strain increment.

The likelihood ratio test of the preceding section can be used to estimate the required number of measurements to achieve a given level of accuracy at a given confidence level. We use the standard deviation of individual range ratio measurements of 5 x 10⁷ as estimated from the present data (Table 3, excluding San Juan). Thus, substituting $\sigma_1 = \sigma_2 = 10^{-6}$ and $|\mu_1 - \mu_2| = 10^{-7}$ into eq. (18), we find that $n_1 = n_2 = 200$ will give t = 1.99, which exceeds the value of t distribution, 1.97, for 198 degrees of freedom at 95% confidence level. Thus a variation in the range ratio of 10^{-7} found by averaging 200 ratio measurements is significant at 95% level of confidence.

At a rate of one measurement every hour, it will take slightly more than a week to complete this many measurements. Two such series of

measurements one year apart is sufficient to determine the shear strain increment in southern California.

For a given set of t and o's, n's are approximately inversely proportional to the square of the difference in μ 's in equation (18). Thus doubling the measurement interval, thereby doubling the expected ratio variations, approximately quarters the required number of measurements. For example, a pair of 50-measurement sets two years apart will give the shear strain rate in southern California.

V. CONCLUSIONS AND RECOMMENDATIONS

Conclusions

Even though the field experiment we performed during this contract was quite limited compared with our original plan, we obtained several interesting and important results. The following is a list of conclusions drawn from these results:

- 1. The increment of the ratio of the length of a survey line to the average of several survey lines in a region is directly related to the incremental shear strain in the region. Thus, the shear strain rate can be calculated from observations of temporal variations in such ratios.
- 2. Using the TLRS, the time-of-flight ratios could be determined to an accuracy (one standard deviation) of 1 x 10⁻⁷ by averaging measurements over a four day period. This accuracy was obtained without using any atmospheric corrections at all. No improvement was obtained when atmospheric corrections based on end-point measurements were applied.
- 3. A calculation using a hypothetical data simulating the observed strain field in southern California indicates that two sets of TLRS ratio measurements separated by one to two years will be sufficient to determine the direction and rate of shear strain in the region.
- 4. Thus relative lateration using the TLRS has been demonstrated to be a good method for monitoring the regional shear strain field around satellite ranging stations. The TLRS operates successfully over long distances. The ratio method is extremely economical. It requires no environmental measuremers and can be performed with small unattended retroreflectors distributed over a wide area. Thus these techniques greatly surpass the capability of conventional EDM techniques.

Recommendations

1. The results of the present experiment are thus very encouraging. However, they are based on only one experiment. Before this technique is put to a practical use, further demonstration is needed to confirm the above results. Therefore, it is recommended that this feasibility study be continued at least to include reoccupation of the Mt. Wilson site and two measurements at another properly selected site, preferably with a different meteorological environment.

- 2. Relative lateration is not limited to the data taken by the TLRS. The data reduction procedure used in the present study can be applied to other data from distance measurements. Therefore, it is recommended that we reanalyze some of existing ranging data to see if improvements in determination of shear strain rate can be achieved. This can be done without further field measurements.
- 3. Additional feasibility test measurements similar to the Mt. Wilson experiment may be obtained from fixed satellite ranging stations. It is therefore, recommended that this possibility be examined.
- 4. Horizontal ranging to distant targets on the ground does not require all the sophistication of the TLRS system. Therefore, when the capability of the present technique is fully demonstrated, a smaller, more portable single-photon ranging unit should be developed for this purpose.
- 5. Finally, the technology is advancing in other fields also. Such techniques as miniature interferometer terminals [Counselman and Shapiro, 1979] may someday be more useful in surveys of regional extent. Therefore, development in these other techniques should be reviewed while developing the present technique.

Acknowledgements

The initial data reduction of the Mt. Wilson experiment was done in the Astronomy Department of the University of Texas at Austin. We are grateful to Dr. Eric C. Silverberg for supplying us the processed data on a computer tape. Dr. Cliff Frohlich kindly reviewed a draft of this report, his constructive comments are greatly appreciated.

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APPENDIX

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ORIGINAL PAGE 19 OF POOR QUALITY

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ដើ ។	15 15 15 15 15 15 15 15 15 15 15 15 15 1	80 8 80 8		698.1		16 10	11.	6		24 83 48 88 24 83 28 88	2 2 3 5 5	741.00	9.253.8 8.8679
			7 7	200	932		Niguel		•	20	10.2	741.8	8.8679
				98	.290		,			83 32		741.8	8.5418
	a)		•	98	.667	83 28	13	738.8	8.8679	69 38		741.1	9.2658
	53			650.1	792.	10 32	12	38	•	10 30		741.1	8.5418
	_	Ø		89	.226	11 30	12	38		11 39		741.1	8. ZZb3
	C)	<u> </u>	•	88	.540	12 30	= :	38		12 31 F	٠	741.1	8,7265 7,6216
	23 00	ତ । ପ	16.1	689.3	wi i	13 31	Ξ,	ž, š				- 45 5 65 5 65 5 65	n (1
	<u> </u>	<u></u>	•	<u> </u>	.797	14 36	4.	χ, έ	9.9151	14 51 15 25		7.48.4	വെ
		•	San Bodro	Ç <u>1</u>		נו א	ָ פַּ פַּ	748.3	18.6688	16 17	10.6	741.1	N
		,) 		17 30	9	8	3	17 28		741.1	c)
22	10 2	1 88	14.7	718.6	9.9131	31	7	40.	333	18 17		741.1	.91
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Table A2. (continued)

ORIGINAL PAGE IS OF POOR QUALITY

v maHg	4.3764	4.5311	4.1783	4.5811	4.8343 5.8648	7.6253 5.8176	6.4397	7.8645 7.2012 6.6937	5.8643	4.7695
P moMg	622.6 622.6 622.6	621.8 621.8 621.8	621.1 621.1 621.1	621.1 621.1 621.1	621.1 621.8 621.8 621.8	621.1 621.1 621.1 620.3	620.3 620.3 620.3 619.6 619.6	619.6 619.6 619.6 619.6 619.6	612.6 613.6 613.6 619.6	618.8 618.8 618.8 618.8
T deg C	1.8	. α ιż	ω 4 ω		ห พ พ เม ห ณ 4 ต	5.6.2	~ ~ @ ~ ~ @ いいのいの	88 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	. N 4 4 W W 4 0 W 0 0 0 V	0.01010101 81-41181
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V mmHg	5.3762		7.3403	6,2908	7.8645	6.5400	5.4836	4.9647	5.3762	4.3764
P mmHg	623.3 623.3 624.1	624.1 624.1 624.1	624.1 624.1 623.3	622.6 622.6 622.6	622.6 623.3 621.8 621.8	621.8 622.6 622.6 622.6	622.6 622.6 622.6 622.6 622.6	621.8 621.1 621.1 621.1 621.1	621.1 621.1 622.6 622.6 621.8	621.8 621.8 621.8 621.8 622.6
T deg C	0.01 8.03 8.03	ක් සි සි සි සි සි සි ස	13.0 0.0 0.0	13.6 11.6 10.3	11.00 0.00 0.00 0.00 0.00	8277	, ഗ.	アケケの4.c. なかのの1.	4 4 4 0 0 4 4 0 0 - 0 4	2.4.00.00.00.00.00.00.00.00.00.00.00.00.0
Time dy hr mn sc	4 8 8 8 8 1	22 16 45 88 22 17 38 88 22 18 88 88	8 1 5 8 6 8 8	21 60 22 60 22 45	22 23 88 68 22 23 45 88 23 88 88 88 23 88 45 88	61, 66 62, 68 62, 45 63, 66	65 66 66 66 66 66 66 66 66 66 66 66 66 6	0 4 4 0 0 0 0 0 0 0 0 0 0 0 0	12 98 12 45 19 38 20 88 21 88	23 23 66 68 24 66 66 66 24 61 45 66 24 62 45 68 24 63 45 66 24 64 45 66
v mmHg	i ui za na	เหต	שייע	14 V V		മ്പമ്പ		9.9131 10.6688 10.2850 9.9131	5.5710	
P mmHg	1 4 10 10 1	738.1 738.1 738.1	ווס וס וכ	333	35 35	322		33333	333333	622.6 623.3 623.3 623.3 623.3
deg C	14.2 16.0	15.55 15.4	9.4	0.00	ច្ច ភ ~ ភ 4		11.1 12.3 14.1 15.0	16.9 15.3 14.4 antiag	1	18.2 18.3 18.3 18.6 18.6
Time dy hr mn sc	19 15 28 23 21 48	22 38 23 38 88 27	01 15 02 27 08 27	64 23 65 48 66 25	87 28 89 23 89 29	11 35 12 29 13 30	25 15 38 08 25 15 17 08 25 17 15 88 25 18 16 08 25 28 28 28 88	21 24 22 26 23 26 69 26	89 88 18 88 10 48 18 46 11 88	22 12 00 00 22 12 42 00 22 12 46 00 22 12 50 00 22 13 60 00 22 14 00 00

Table A2. (continued)

ORIGINAL PAGE FOOF POOR QUALITY

Time	-	α.	>	Time	-	٥.	>	Time		
dy hr mn sc	deg C	Ħ	mmHg	dy hr mn sc	deg C	mirHg	maHg	dy hr mn sc	deg C maya	g mr.Hg
02 43	2.3	618.8		26 88 28 88	ហ	617.3	6.1694	04 45	וֹנָיו	3 8.3828
82 51	2.4	618.8	4.7315	•				000 000 000 000 000 000 000 000 000 00	ni L	3 K. 7263
25 03 00 00	ю. 191	618.8	•		San Juan	£ !		24 85 38 88 24 86 37 88	20 00 20 02 20 02 20 03	3 8.87.24 3 7.8313
45 to	_ r	618.8		22 BB BB BB BB		-	9.8638	82 83	· [~	3 7.6848
27 72		618.0	4.5811	69 63	15.8	717.8		<u> </u>	.8 718	~
04 75 04 25	· G	618.8	•	10 02	•		•	11 01	.0 718	.6 7.3403
22 23	, L	618.8	••	10 58			•		٠. م	6 7.3463
65 25	ω.	618.8		1	16.0	7	9.6957	13 10	2	6 7.1737
96 15	1.8	618.8		12 59	16.5	717.8	10.6299	4 14	.5 717	r- 1
06 23	2.1	618.8	₹.5811	14 26		•	11.8653	15 02	.0 717	.8 7.6645
06 26	2.0	618.8		22 15 83 88	16.0	718.6	7.8618	24 15 03 08	8.8 718.	7.4334
06 35	6.1	618.8		16 85 15	ກຸ	7.95 2.65 2.65	7.8815 0.4643	17 22	- r	- 00 1 40
96 37	6.1	618.8		79 VI			0.1045	10 00	- ^	8 9.7318
06 50	2.1	618.8		200	יי ע מייע מייע	7.7. 0.0.0	0.0410 .0.0550	9 5	. r-	9,8402
07 15		618.8		2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	ם נ ניני	ייי טייי	19.2038	200	11.	31.52.9
07 25	8	618.1		79 79 70 70 70	7.77	713.6	18.2838	ין מין מין	716	ነወ
68 15	o,	618.1	4.3764	23 GG	5 u	717.1	10.0000	22 26	٠.	, ,
08 20	N.	618.1		מ מ מ מ מ מ מ מ מ מ מ מ מ מ מ מ מ מ מ		717.1	11.4747	07 22 20 00	717	6250 P
09 24		618.1		23 23 26 26 27	, 10 10 10 10 10 10 10 10 10 10 10 10 10	717.0	11.3316		715	,
89 2G	1:1	617.5				717.0	10.0313	200) (C)
10 15		617.3				717.0	0 9121	36 78	7 7 5) t~
97 G		617.3	7 110	200	•	710.0	9.0121	1 10		6 7,6258
18 25	æ (√ ₽	4.2774	3 5	2.0	718.6	0.0101	74 35	715	.6 7.3518
00 00 00 00 00 00 00 00 00 00 00 00 00	0.	617.5		100 100 100 100 100 100 100 100 100 100	•	710.0	9.7318		9.5 715.6	i in
2 2 2 3 3 6 3 6 7		517.5			•		9.5532	96 35		~
27		17.0		00 02 04 07 08			9.5532	87 30	ım	~
3 5		• •		88 82			9.3881	88 32	3 714	_
12 15			3.8100	88 68 88 88	12.0	716.3	9.0358	69 38	3 713	G
12 20	'n	616.6	4.3764	10	10.5	717.1	9.2392	18 31	713	.3 6.1935
12 23	4	616.6		10 56	11.0		9.5179	11 30	N- 1	3 6.6578
13 15	9.			12 03	11.5	717.8	9.4475	12 31	N	5 6.7578
13 20	ស			18 00	•	719.3	9.1026	13 35	t	
14 20	i,	17.	4.3764	19 88	15.5	•	9.5532	14 29 15 29	t	1 C
15 20	1.9	618.1		20 02	•	•	18.2858	15 52 52 51	7 7 7	Ċ
16 28	4.7	618.1		21 00			•	16 52	מון מ	000000 00000 00000
17 20	5	$\boldsymbol{\omega}$		22 10	14.5	•	•	3 :		0
18 20	6.2	18	6.0500	23 05	12.9	•	9.2858	18 55	71.7 to	o o
19 20	7.7	618.1		00 00	٠		9.2826	85 E		•
	8.3	~ :	6.2908	000	11.5 2.5	718.6	9.8358	25 28 35 88 75 24 45 86	18.8 (13	5 6.5626 5 7 9218
21 20	9.4	~		05 66	•	719.5	g. 55.23	ין קר	, ,	•
22	8.8	617.3		63	တ တ	719.3	. •	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	20.02	2 2 20 2 2
23 20		~		04 00	လ ရာ	719.3	8.0724	3	. 1	

Table A2. (continued)

ORIGINAL PAGE 19 OF POOR QUALITY.

	! ! ! ! ! ! ! ! ! ! ! ! ! ! ! ! ! ! !					Timo	¢
dy hr mm sc	c	dy hr mn sc	=	dy hr nn sc	:	dy hr mn sc	:
		24 13 54 08	1.08825888	01 03	1.00024528	24 69 86 66	1.63027267
Mt. Wilso	son	14 28		27 21 13 23	1.00024919	19 GB	1.03327286
: t	!	.14 54	1.80825888	•		11 00	1.66927296
22 12 08 01 1.	.00024386	24 15 53 54	1.66825848 1.66627959	Castro	2 !	24 12 69 88 24 13 68 88	1.88327296 1.88327296
15 80 34 1.	. 66624413 . 66674414	17 53	1.88824796	22 89 18 88	1.00026392	14 89	1.09027325
15 59 57	.88824414		1.00024679	18 88	1.00026383	15 99	1.88827365
16 31 16 1.	.00024327	19 55	1.60024679	Ξ	1.00025443	16 83	1.68827315
16 57 27 1.	.00024414	20 49	1.00024558	12 00	1.00026465	8	1.05827198
17 59 44 1.	.00024272	21 54	1.00024411	13 00	1.00026457	18 9	1.89927953
19 14 19 1.	0002407	22 27	1.60024411		1.88826493	19 gg	1.68836938
21 57 23 1	.00024183	22 22	1.06924411	15 00	1.00026502	26 62	1.60926747
22 25 11 1	.00024129	23 22	1.00024588	16 00	1.00026447	21 20 20 20 20 20 20 20 20 20 20 20 20 20	1. Justes 43
23 02 15 1	.00024183	00 34 1	1.00024499	17 88	1.00026459	22.02	1.83825738
60 60 56 1	.86824269	02 00 1	1.63024766	33 S	1.88826376	25 C	1. Man.abold
00 28 23 1	.00024355	63 63	1.00024857	99 61 61	1.00025234		1. Backarra .
01 03 14 1	.00024355	63 22	1.06624948	200	1.08625994	⊒ (1.00027123
62 61 37	.00024442	04 32	1.00024857	22 21 88 88	1.00025225 1.00025230	20 82 86 86 50 86 86 86	1.05057167
62 24 63 60 61 63	. 88824442	מט מט מט מט מט מט מט	1.86884837	77 60	1.00020320	200	1 69307725A
62 48 26 64 17 66	.88824329		1.80664837 1.66834837	200	1.0002623		1.65827215
20 10 20 20 20 20 20 20 20 20 20 20 20 20 20	.00024323	20 70	1.00024031			90	1,68322187
84 33 34 85 57 22	1.00024616	90 20	1.08824313 1.68874887		1.88826522	92 GB	1.88827256
86 37 R7	. 66624618	69	1.05324796	03 00	1.00026764	25 68 89 88	1,58827166
06 55 29 1.	.00024529	10 28		84 88		69 63	1.80827175
07 52 58 1	.00024550	11 69	1.85824678	02 00	1.00026634	00	1.96927195
08 20 32 1	.09024590	11 59	1.00024859	0e 60	1.00026643	11 39	1.80827195
09 54 34 1.	.00024590	12 34	1.05024768	16 00	1.00027186	8	1.60327254
09 56 13 1	.00024679	13 57	1.00024678	8	1.00027034	13 90	1.00927148
10 29 17 1	.00024679	14 26	1.05024768	18 88 8	1.00026863	5 14 68 14 68	1.88327285
10 55 37 1	.00024736	14 56	1.00024796	3 19 68 3 19 68	1.00926684	15 25 26 26	1.60627244
04 06 03 1.	.88025848	15 55	1.00024738	2 Z Z	1.8882882		1.88887.455 1.88867.4558
04 22 39 1.	.00025132	16 36	1.00624549	22 22 68 66	1.88826327 1.88826756	3 E	1.08/02/505
84 33 23 I	. 000070040 000070070	 S &	1 14500001	77 20 27 27 29 27 27 27 27 27 27 27 27 27 27 27 27 27	1.000203	19 88	1.05526802
85 34 27 1	.00823040	5 6	1.06624352	4 60 60	1.00026965	5 29 68	1.60326559
4 67 64 16	.00025040	19 55		4 01 00	1.00027119	21 69	1.86625171
4 07 51 41 1		20 31	1.00024265	4 02 00	1.00027255	22 98	1.03025355
4 68 31 26 1	ın	21 06	1.00024179	63 60	1.90027264	က အ (1.83326188
4 09 46 20 1	. 68925848	21 54	1.00024238	94 88	1.00027296	26 88 89 86	1.88825348
4 10 28	23.18 1.18	22 32	1.00024152	90	1.00027286	G (8)	Dodo
4 11 00 4	2513	23 82	1.88824238	24 85 88 88	1.00027275)
24 12 34 12 1.	.00025180	00	1.00024352	88	ដូ	22 10 21 89	1.88927425

Group Index of Refraction at End Points

Table A3.

ORIGINAL PAGE IS OF POOR QUALITY

•	Table A3.	(continued)					
Time dy hr mn sc	c	Time dy hr mn sc	c	Time dy hr nn sc	E .	Time dy hr nn sc	С
12	1.00027576	13	1.00027924	19 15	1.00028302		1.85824280
14 20	002768	20	1.00027792	24 20 23 00	1.00028969	15 83	1.86824166
16 20	000273	-21 40	1.06628962	21 40	1.00028033	16 99	1.80304214
18 20	000270	22 30	1.08028137	30	1.60028591	16 35	1.85024214
22 00	000270	23 37	1.06920385	23 30	1.00028101	17 88	1.66324214
0	1.00027733	60 38	1.00028384	00 27	1.00028327	13 00	1.69824146
16 20	000277	01	1.66628534	01 15	1.00028536	18 45	1.66823968
18 20	1.00627731	82 32	1.00028464	02 27	1.00028698	19 GB	1.65823960
20 30	1.00027509	83	1.88828534	83 38	1.09928898	20 83	1.86823971
24 80 28 88	1.06027811	84 46	1.00028524	53	1.00028851	21 69	1.88823983
04 20	1.00028055	85 38	1.00028534	05 40	1.00028728	55 80 1	1.63924828
96 10	1.89825825	86 32	1.00028543	06 25	1.00028717	22 45	1.68524138
06 45	1.00028035	02 20	1.60028533	07 28	1.00028758	23 99	1.08324371
68 10	1.00028038	08 30	1.00928228	08 23	1.00028711	23 45	1.66324251
10 10	1.68028107	69 29	1.06928393	69 53	1.00028720	99 99 9	1.63324193
10 45	1.00028117	10 35	1.86828423	10 38	1.00028710	99 451	1.83824356
12 10	1.00028077	11 38	1.66828423	11 35	1.00028760	61 88	1.63324796
	1.00028097	12 49	1.00028413	12 29	1.00028751	675 GE	1.60624597
	1.06828127	17.48	1.86828469	13 30	1.00028761	62 52	1.88324554
02 10	1,88827927	18 34	1.88828429	14 25	1.00028843	83. GB	1.506543588
02 52	1.06627927	19 38	1.09028429	15 38	1.888598		1.68823484
04 10	1.00027927	20 40	1.60628128	16 17	1.00028495	14 45 45 45 45 45 45 45 45 45 45 45 45 45	1.00025584
66 10	1.00027998	21 33	1.60028157	17 15	1.80828486	3 6 3 6	
06 46	1.00028019	22 30	1.60928205	16	1.00028198	5 5 5 6 7 7	1.60000000
63 10	1.00028099	23 26	1.00028137	19 24	1.69028843	3 G	1.60% 4454
10 10	1.00027999	80 30	1.86828452	26 26	1.88827929))) (a)	
10 5	1.60027999	01 46	1.00028679	21 24	1.000023	ည က က က	. doubthor
12 10	1.08028999	82 48	1.03828718	22 28	1.88827812		1.80004400
14 10	ရွ	03 20	1.00028699	23 26	1.88827966		1.00000000
14 22	1.00028119	04 35	1.99828718	26 gg 28 gg	1.00025384	10 C	
16 1	1.00027770	8	1.00028750			25 18 89 50	1.000000000000000000000000000000000000
		9 2 50	1.90923759	sant iago	යල්ට		1. SCORVACION .
Niguel	[a]	07 35	1.88828768	' (20 12 00 00	**************************************
		08 32	1.9882811	200	1.00024038	7 5	1.00025000
2 09 28	1.00028344	89 38	1.93828713	39 39 5	1.88824157	5 5 6 6 7	
2 10 32	1.60028413	10 30	1.80028835	16 48	1.68824174	בר ה בר ה בר ה	1.888844301
2 11 30	1.00028463	11 33	1.80028986	18 6	1.00024166	21 22	. 00000000
2 12 30	-	12 31	1.00028917	8	1.00023980	35 CE	I.uddzeszy
2 13 31	1.00028513	13 33	1.00028927	12 68 5	1.88824159	36	2 000000000. 2 0000000000000000000000000
2 14 30	.000282	14 31	1.00028958	12 42	1.08824148	65. G	1.6202020
2 15 41	1.00028362	ស	1.00028793	9 년 연기	1.00024157	-	Province.
22 16 36 00	1.69928224	16 17	1.06028693	12 5g	1.66624157	3 t	1.0200000001 020000000001
2 17 30	1.06928187	24 17 28 88	1.8885592	00 00 F. CC	1.00024163		Manager -
2 18	1.86828828	18 17	1.88828432	14 80	1.88824288		1.3302.4501
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ORIGINAL PAGE 19 OF POOR QUALITY

Tima	i 6 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			Time	: : : : : : : : : :	Time	- c
dy hr mn sc	:	dy hr nn sc		dy hr masc		dy hr nn sc	
24 65 45 60	1.00024949	02 43	1.38324679	26 88 28 88	1.00024337	12	1.63928856
06 25	1.80824948	02 51	1,00024570			සිය වන	1,82828856
06 48	1.06024931	99 29	1,88024579	San Ju	Juan	(1) (2) (3)	1.60328185
67 20 20 20 20 20 20 20 20 20 20 20 20 20		63.	1.08024733		1 00007771	2 0	1,6552,555
200	1.00024953 1.0002A953	22 CB 84 15	1.300247.31	89 83	8827	10 85	1.20027979
2 2	1.00024907		1.90024796	10 02	1.00027338	11 61	1.66626.379
11 20	1.68024943	05 15	1.98024842	10	1.00027385	12 53	1.69225979
12 20		05 25	1.06624815	11 58	1.00027271	13 10	1.86228379
13 20	1.00024907	86 15	1.00024724	12 59	1.86027224	14 88	1.68828831
14 20	1.00024816	86 23	1.00324697	14 26	1.00027290	15 82	1.60826148
4 15 20	1.00024690	96 26	1.06024705	15 83	1.00027302	16 65	1.66628979
16 20	1.00024780	06 35	1.00024715	22 16 85 88	1.88827575 . 88827237	24 16 55 88 24 17 71 66	1.6862515
17 28	1.88824692	86 SY	1.86824713	<u>-</u> 0	1.00027070	12 24	1 66821538
	1.88874788 1.88874788	20 21 CD 2C	1.00024636	0 0	1 88825458) E	1.08027339
		2 6 2 7 2 12	1 06004687	200	1.00026729	i, G	1.09327168
700	1 00024212	2 6 2 7	1,00024865	21 88	1.68826873	21 36	1.68027262
20 05	1.83824417	88 28 88 28	1.06624841	22 00	1.00027011	22 38	1.98327188
100	1.00024316	88 24	1.00824858	22 23 66 68	1.88827884	23 39	1.09827336
20 40		69 20	1.63824727	98 B8	1.00027366	09 30	1.68327569
50	1.00024281	10	1.00024655	3 01 00	1.00027482	01 33	1.69927813
21 60			1.00024664	02 09	1.09927654	82 36	1.66827912
21 05	1.88824253	10 23	1.00024664	83 88	1.00027782	නි දිනි දිනි	1.83327961
21 15	1.86824219	10 35	1.80824682	<u> </u>	1.69927685	1 33	1,65627912
<u> </u>	1.00024150	10 38	1.09024673	85 68	1.00027733	600 1 60 1 60 1 60 1 60	1.65527815
22 15	1.60824133	10 58	1.00024664	00 00 00 00	1.00027751	60 60 60 60 60 60 60 60 60 60 60 60 60 6	1.65627942
22 20	1.00024142	11 15	1.66024638	3 07 00	1.00027703	87 87 80 80 80 80 80 80 80 80 80 80 80 80 80	1.60627983
22 23	1.68624898	11 20	1.00024638	88 62	1.03027627	N 6	1.68827883
22 32	1.06024228	12 15	1.00624726	60	1.86827596	90 90 68 67 90 90 90 90	1.09027773
22 35	1.00024253	12 28	1.00024772		1.888ZZZZZ	3 6	1.0000001
24 22 45 88	1.00024585	25 12 23 88	1.88824762 1.68834773		1.00027.01 1.00027703	12 22	1.65527921
ק ק ק	1 06024420	7 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1. FRB24782	18 89	1.00027663	13 35	1.69923592
5 60 15	1.08024502	14 20	1.00024854	91	1.00027376	14 25	1.86028802
5 60 20		15 20	1.00024687	28	1.00027349	15 32	1.66927802
5 60 25	1.00024578	16 2	1.86824439	21 00	1.66627588	16 32	1.88827666
5 68 29	1.00024537	17 20	1.00024396	22 10	1.00027414	17 33	1.65827894
5 01 15	1.00024635	18 20	1.00024335	23 05	1.00027654	18 33	1.88827389
01 29	1.00024644	19 20	1.00024178	88 88	1.00027733	19 30 19 1	1.63027178
5 02 15	1.00024670	20 20	1.80824895	01 00	1.00027733	28 35	1.65626914
5 02 27	1.88824679	1 20	1.08824881	92 98 01 01	1.88827957	21.45	1.88825842
5 02 32		22 28	1.60024052	E 2 E E E	1.66628666	25 22 55 88 27 56 35 88	1. COUCACTES
20	1.88824679	23 2	1.66624224	24 84 88 88 	1.00026006	2	1.0000.00

Table A3. (continued)

Ratios
of-F1 ight
f. Time-of
Table A4.

Castro	San Pedr	2	Niguel		Sant iago		San Juan	
dy hr mm sc 1,184+	dy hr mn sc	0.948+	dy him sc	1.364+	dy hr mm sc 1	.210+	dy hr masc	0.735+
19 PC PU BI	18 18	870439	24 84 28 32	486752	16 41 41	567622	18 51	646886
22 12 62 41 915176	22 12 21 52	871345	9	487793	43 11	567332	2	645954
14 05 43 91	14 15 03	870661	8	485966	14 34 32	566961	14 33	647619
88 89 48 91	18 23 39	871637	68 13	486255	22 49 55	566262	18 55 83	643189
62 69 25 91	84 28 48	870474	10 15	486647	69 41 66	566832	22 54 58	5-8337
06 85 83 91	96 16	871245	10 40	436857	04 33 04	566222	05 47	64, 3, 5
06 08 46 91	05 51 04	878698	14	486632	39 34	565878	47 14	647283
06 58 17 91	08 69 40	871172	14 40	486914	68 39 44	565151	84 45 14	648393
16 60 19 91	19 19 93	870674	16 18	486108	04 40 08	566802	8 8	647595
89 59 54 91	10 45	820203	02 21	486409	86 25 43	566603	96 36	643839
10 55 03 91	12 10	878911	02 43	485806	06 41 31	565969	88 23	648657
12 88 58 91	14 11 38	870981	10	486836	08 22 17	566526	12 25	648343
14 01 34 91	14 45 60	870330	10 41	486512	10 21 13	566338	14 23	640772
14 55 85 91	02 13	870356			10 35 15	566618	14 38	649358
16 81 83 91	02 53	871173			12 20 63	566316	16 25	549338
18 00 06 91	64 69	22698			14 20 03	564893	82 33	643673
18 57 39 91	96 19	870052			14 35 11	564750	62 36	640634
22 44 42 91	06 45	870209			16 29 53	565410	04 25 60	658826
02 05 25 91	03 11	871274			29 18 64	565626	85 32 42	648312
02 59 84 91	10 10	871390			88 25 13	565889	13	647559
64 66 52 91	10 45	870584			82 29 38	565803	13 23	647534
86 81 17 91	12 10	871738			02 39 51	565623	10 36	647.775
66 55 21 91	14 15 11	871417			04 22 37	566544	12 25	648233
68 64 49 91	14 48 42	871316			86 22 51	566702	18 31	647344
10 01 53 91					86 35 48	566195	20 25 59	648656
10 55 08 91					03 20 32	566493	22 28 53	645712
12 04 05 9	,				10 20 08	566462	60 32 59	648933
14 51 54 91	•				10 35 11	566862		
16 00 28 91	~				12 21 02	565658		
18 88 19 91					14 21 41	566251		
18 59 43 91					14 35 02	566204		
20 67 42 91					16 20 06	567037		
20 37 24 91					18 39 42	556858		
20 49 58 914					28 28 59	556413		
22 51 57 912		,				565985		
88 85 27 912					00 26 53	565337		
							}	

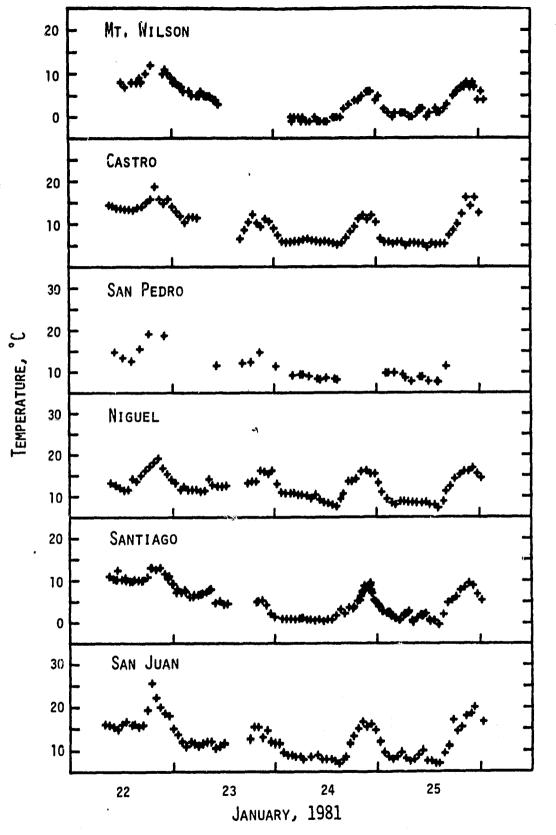


Fig. Al. Observed atmospheric temperature.

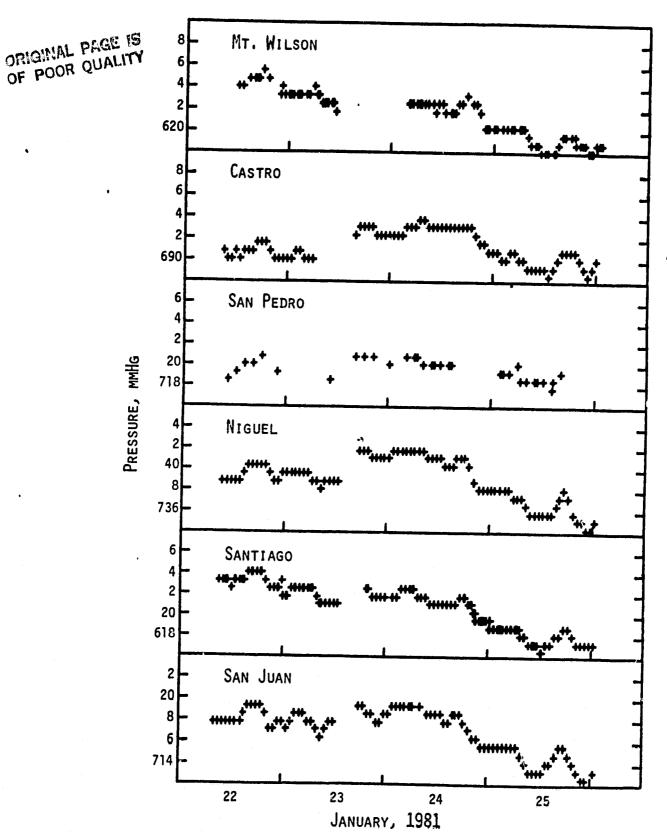


Fig. A2. Observed atmospheric pressure.

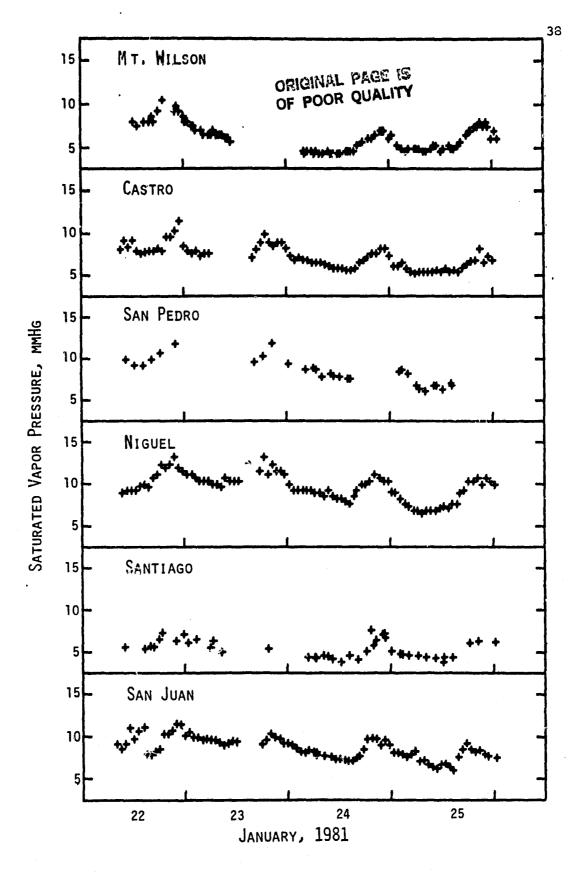


Fig. A3. Saturated vapor pressure.

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